

Appendix A:

Equations of motion in General Three-Body Problem

$$\ddot{\mathbf{r}} = \left[\begin{array}{c} \ddot{x} \\ \ddot{y} \\ \ddot{x} \\ \ddot{y} \\ \ddot{x} \\ \ddot{y} \end{array} \right] = \left[\begin{array}{c} \ddot{p}_1 \\ \ddot{p}_2 \\ \ddot{p}_3 \end{array} \right]$$

$$\ddot{p}_1 = \frac{M_2}{\|x_2 - x_1\|^3} (x_2 - x_1) + \frac{M_3}{\|x_3 - x_1\|^3} (x_3 - x_1)$$

$$\ddot{p}_2 = \frac{M_1}{\|x_1 - x_2\|^3} (x_1 - x_2) + \frac{M_3}{\|x_3 - x_2\|^3} (x_3 - x_2)$$

$$\ddot{p}_3 = \frac{M_1}{\|x_1 - x_3\|^3} (x_1 - x_3) + \frac{M_2}{\|x_2 - x_3\|^3} (x_2 - x_3)$$

Appendix B:

Derivation of the Lagrange - Jacobi identity: *

In the center of mass coordinate system, the moment of inertia of the three-body system is:

$$I = m_1 r_1^2 + m_2 r_2^2 + m_3 r_3^2$$

Or in the Lagrangian form:

$$I = \frac{m_1 m_2 m_3}{M} \left(\frac{r_{21}^2}{m_3} + \frac{r_{13}^2}{m_2} + \frac{r_{32}^2}{m_1} \right)$$

where:

$$M = m_1 + m_2 + m_3$$

$$r_{21} = \|\vec{r}_2 - \vec{r}_1\|$$

$$r_{13} = \|\vec{r}_1 - \vec{r}_3\|$$

$$r_{32} = \|\vec{r}_3 - \vec{r}_2\|$$

Differentiating I twice with respect to time gives

$$\dot{I} = 2m_1 \vec{r}_1 \cdot \dot{\vec{r}}_1 + 2m_2 \vec{r}_2 \cdot \dot{\vec{r}}_2 + 2m_3 \vec{r}_3 \cdot \dot{\vec{r}}_3$$

$$\ddot{I} = 2m_1 (v_1^2 + \vec{r}_1 \cdot \ddot{\vec{r}}_1) + 2m_2 (v_2^2 + \vec{r}_2 \cdot \ddot{\vec{r}}_2) + 2m_3 (v_3^2 + \vec{r}_3 \cdot \ddot{\vec{r}}_3)$$

$$\frac{1}{2} \ddot{I} = 2T + V = 2E - V$$

$$T = \frac{1}{2} \sum_{i=1}^n m_i v_i^2, \quad E = T + V, \quad V = -G \sum_{j=1}^n \sum_{k=j+1}^n \frac{m_j m_k}{r_{jk}}$$

From Valtanen p32 - p33

Appendix C:

Translate Equations of Motion to Rotating Inertial Frame:

To go to the rotating inertial reference frame we first scale the problem to depend on a single parameter, μ .

- Mass of planet = μ
- Mass of sun = $1 - \mu$

$$\mu = \frac{\text{mass of planet}}{\text{total mass}} = \frac{m}{m+M}$$

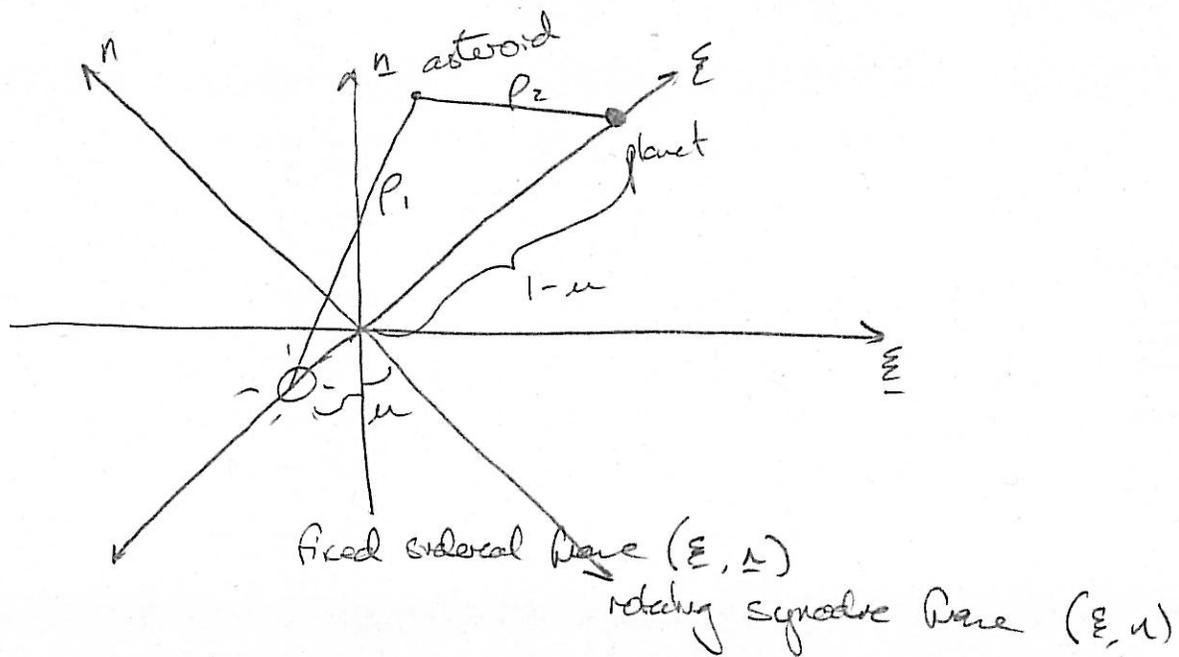
- distance between primaries (planet and sun) = 1
 - distance from center of mass to planet = $1 - \mu$
 - distance from center of mass to sun = μ
- unit of time chosen so that mean motion of primaries is $n = 1$.
hence mean anomaly = time
- From these it follows that the gravitational constant, $G = 1$.

To describe the rotating inertial frame, we first start in the fixed sidereal frame: $\underline{\xi}, \underline{n}$. In this frame the equations of motion are:

$$\text{Sun} \begin{cases} \underline{\xi} = -\mu \cos t \\ \underline{n} = -\mu \sin t \end{cases}$$

$$\text{Planet} \begin{cases} \underline{\xi} = (1-\mu) \cos t \\ \underline{n} = (1-\mu) \sin t \end{cases}$$

But, we want to translate to the rotating, synodic frame $\underline{\xi}, \underline{n}$ where the primaries are stationary.



The Hamiltonian of the asteroid in the fixed frame is:

$$\underline{H} = \frac{1}{2} (p_{E'}^2 + p_{n'}^2) - \frac{1-u}{p_1} - \frac{u}{p_2}$$

Where $p_1 = ((E + u \cos t)^2 + (n + u \sin t)^2)^{\frac{1}{2}}$

$$p_2 = ((E - (1-u) \cos t)^2 + (n - (1-u) \sin t)^2)^{\frac{1}{2}}$$

The fixed frame \rightarrow rotating frame are related by the following equations:

$$E' = E \cos t - n \sin t$$

$$n' = E \sin t + n \cos t$$

The transformation can be accomplished using a generating function:

$$F = F(p_{E'}, p_{n'}, E, n)$$

$$= -(E \cos t - n \sin t) p_{E'} - (E \sin t + n \cos t) p_{n'}$$

The new momenta are:

$$P_{\xi} = -\frac{\partial F}{\partial \xi} = p_{\xi} \cos t + p_n \sin t$$

$$P_n = -\frac{\partial F}{\partial n} = -p_{\xi} \sin t + p_n \cos t$$

The new Hamiltonian is:

$$\begin{aligned} H = \underline{H} + \frac{\partial F}{\partial t} &= \frac{1}{2} (P_{\xi}^2 + P_n^2) - \frac{1-\mu}{\rho_1} - \frac{\mu}{\rho_2} \\ &+ \xi p_{\xi} \sin t + n p_{\xi} \cos t + \xi p_n \cos t + n p_n \sin t \\ &= \frac{1}{2} (P_{\xi}^2 + P_n^2) - \frac{1-\mu}{\rho_1} - \frac{\mu}{\rho_2} + \xi(-P_n) + n P_{\xi} \end{aligned}$$

where:

$$\rho_1^2 = (\xi + \mu)^2 + n^2$$

$$\rho_2^2 = (\xi - (1-\mu))^2 + n^2$$

The equations of motion in the synodic frame are now:

$$\dot{\xi} = \frac{\partial H}{\partial P_{\xi}} = P_{\xi} + n$$

$$\dot{n} = \frac{\partial H}{\partial P_n} = P_n - \xi$$

$$\dot{P}_{\xi} = -\frac{\partial H}{\partial \xi} = \frac{\partial}{\partial \xi} \left(\frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} \right) + P_n = \frac{\partial}{\partial \xi} \left(\frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} \right) + \dot{n} + \xi$$

$$\dot{P}_n = -\frac{\partial H}{\partial n} = \frac{\partial}{\partial n} \left(\frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} \right) - P_{\xi} = \frac{\partial}{\partial n} \left(\frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} \right) - \dot{\xi} + n$$

$$\dot{P}_{\xi} = \ddot{\xi} - \dot{n}, \quad \dot{P}_n = \ddot{n} + \dot{\xi}$$

So:

$$\ddot{\xi} - \dot{n} = \dot{n} + \xi + \frac{\partial}{\partial \xi} \left(\frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} \right)$$

$$\ddot{n} + \dot{\xi} = -\dot{\xi} + n + \frac{\partial}{\partial n} \left(\frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} \right)$$

These equations simplify to:

$$\ddot{\xi} = 2\dot{\eta} + \xi - (1-\mu)(\xi + \mu) \left[(\xi + \mu)^2 - n^2 \right]^{-3/2} \\ - \mu (\xi - (1-\mu)) \left[(\xi - (1-\mu))^2 + n^2 \right]^{-3/2}$$

$$\ddot{\eta} = -2\dot{\xi} + \dot{\eta} - n(1-\mu) \left[(\xi + \mu)^2 + n^2 \right]^{-3/2} \\ - n\mu \left[(\xi - (1-\mu))^2 + n^2 \right]^{-3/2}$$

These equations form the following system of coupled
two order equations

$$\dot{\xi} = p$$

$$\dot{p} = 2q + \xi - (1-\mu)(\xi + \mu) \left[(\xi + \mu)^2 - n^2 \right]^{-3/2} \\ - \mu (\xi - (1-\mu)) \left[(\xi - (1-\mu))^2 + n^2 \right]^{-3/2}$$

$$\dot{\eta} = q$$

$$\dot{q} = -2p + \eta - n(1-\mu) \left[(\xi + \mu)^2 + n^2 \right]^{-3/2} \\ - n\mu \left[(\xi - (1-\mu))^2 + n^2 \right]^{-3/2}$$

q ?

Appendix D

Lagrange Points: Locations and Stabilities

Locations:

Lagrange points occur where $\dot{\xi} = \dot{\eta} = 0$

This occurs when $\frac{\partial \Omega}{\partial \xi} = \frac{\partial \Omega}{\partial \eta} = 0$ where

$$\Omega = \frac{1}{2} (\xi^2 + \eta^2) + \frac{1-\mu}{r_1} + \frac{\mu}{r_2}$$

but Ω is not H the Hamiltonian - why not?
 r_1 & r_2 defined in Appendix C

because

$$\ddot{\xi} - 2\dot{\eta} = \frac{\partial \Omega}{\partial \xi} \quad \text{and} \quad \ddot{\eta} + 2\dot{\xi} = \frac{\partial \Omega}{\partial \eta}$$

Calculating these derivatives we get a pair of equations for the Lagrange equations

$$\frac{\partial \Omega}{\partial \xi} = \xi - \frac{(1-\mu)(\xi+\mu)}{r_1^3} - \frac{\mu(\xi-(1-\mu))}{r_2^3} = 0 \quad (5.17)$$

$$\frac{\partial \Omega}{\partial \eta} = \eta - \frac{(1-\mu)\eta}{r_1^3} - \frac{\mu\eta}{r_2^3} = 0 \quad (5.18)$$

The second equation has the trivial solution where $\eta = 0$. The first equation then gives:

$$\xi - \frac{(1-\mu)(\xi+\mu)}{[(\xi+\mu)^2]^{3/2}} - \frac{\mu(\xi-(1-\mu))}{[\xi-(1-\mu)]^2}^{-3/2} = 0$$

This has the solution:

$$\xi - \frac{1-u}{(\xi+u)^2} - \frac{u}{(\xi-(1-u))^2} = 0 \quad 1-u < \xi$$

$$\xi - \frac{1-u}{(\xi+u)^2} + \frac{u}{(\xi-(1-u))^2} = 0 \quad -u < \xi < 1-u$$

$$\xi + \frac{1-u}{(\xi+u)^2} + \frac{u}{(\xi-(1-u))^2} = 0 \quad \xi < -u$$

These can be solved numerically.

For $n \neq 0$ we have the second equation

$$1 - \frac{(1-u)}{p_1^3} - \frac{u}{p_2^3} = 0 \quad (5.18)$$

Multiplying by $\xi+u$ and subtracting the first equation (5.17)

$$\frac{u}{p_2^3} - u = 0$$

therefore $p_2 = 1$

Multiplying (5.18) by $\xi-(1-u)$ and subtracting (5.17) we get

$$1-u - \frac{1-u}{p_1^3} = 0$$

therefore $p_1 = 1$

Since $p_1 = 1$ and $p_2 = 1$ then the points are those on the equilateral triangle with the points of the primaries:

$$(\xi - (1-u))^2 + u^2 = 1$$

$$(\xi + u)^2 + u^2 = 1$$

$$\circ \circ \quad \xi = \frac{1}{2} - u, \quad u = \pm \frac{\sqrt{3}}{2}$$

Stability

We can explore the stability perturbing a body near a Lagrangian point.

Let (ξ_0, η_0) be any Lagrangian point and (x, y) the position of the body relative to this point:

$$x = \xi - \xi_0$$

$$y = \eta - \eta_0$$

Close enough to the neighborhood about the Lagrangian point we can linearly approximate the derivatives of Ω

$$\frac{\partial \Omega}{\partial \xi} = \frac{\partial \Omega}{\partial \eta} = 0$$

$$\frac{\partial \Omega}{\partial \xi} \approx x \frac{\partial^2 \Omega}{\partial \xi^2} + y \frac{\partial^2 \Omega}{\partial \xi \partial \eta}$$

$$\frac{\partial \Omega}{\partial \eta} \approx x \frac{\partial^2 \Omega}{\partial \xi \partial \eta} + y \frac{\partial^2 \Omega}{\partial \eta^2}$$

So the equations of motion are

$$\ddot{x} - 2\dot{y} = x \frac{\partial^2 \Omega}{\partial \xi^2} + y \frac{\partial^2 \Omega}{\partial \xi \partial \eta}$$

$$\ddot{y} + 2\dot{x} = x \frac{\partial^2 \Omega}{\partial \xi \partial \eta} + y \frac{\partial^2 \Omega}{\partial \eta^2}$$

Considering the three Lagrangian Points L_1, L_2, L_3 the derivatives of Ω are:

$$\frac{\partial^2 \Omega}{\partial \xi^2} = 1 + 2\alpha$$

$$\frac{\partial^2 \Omega}{\partial \eta^2} = 1 - \alpha$$

$$\frac{\partial^2 \Omega}{\partial \xi \partial \eta} = 0$$

$$\text{where } \alpha = \frac{1-\mu}{\rho_1^3} + \frac{\mu}{\rho_2^3}$$

as $\eta = 0$ for L_1, L_2 and L_3

The equations of motion are now:

$$\ddot{x} - 2\dot{y} = x(1+2\alpha)$$

$$\ddot{y} + 2\dot{x} = y(1-\alpha)$$

We can study their trajectories to determine stability:

$$x = A e^{\omega t}$$

$$y = B e^{\omega t}$$

where A, B , and ω are constants

If $\text{Re}(\omega) \neq 0$ then the solutions grow without limit and are unstable.

If $\text{Re}(\omega) = 0$ then the solutions are stable.

Substituting in the solution and performing some algebra we see:

$$A\omega^2 - 2B\omega = A(1+2\alpha)$$

$$B\omega^2 + 2A\omega = B(1-\alpha)$$

Eliminating A & B and performing more algebra we get

$$\omega^4 + \omega^2(2-\alpha) + (1+2\alpha)(1-\alpha) = 0$$

If $\text{Re}(\omega) = 0$ then there must be two negative solutions for ω^2 .

The product of these roots must be positive, therefore:

$$(1+2\alpha)(1-\alpha) > 0$$

therefore: $\alpha < 1$

But the Lagrangian points must satisfy (5.17) from Appendix C

$$\frac{\partial \Omega}{\partial \xi} = \xi - \frac{(1-u)(\xi+u)}{\rho_1^3} - \frac{u(\xi-(1-u))}{\rho_2^2} = 0$$

Rearranging the terms

$$\xi - \xi \left[\frac{1-u}{\rho_1^3} + \frac{u}{\rho_2^2} \right] - \frac{u(1-u)}{\rho_1^3} + \frac{u(1-u)}{\rho_2^3} = 0$$

$$\text{or } \varepsilon(1-u) - u(1-u) \left[\frac{1}{\rho_1^3} - \frac{1}{\rho_2^3} \right] = 0$$

from which we see

$$1-u = \frac{u(1-u)}{\varepsilon} \left[\frac{1}{\rho_1^3} - \frac{1}{\rho_2^3} \right]$$

Therefore for all Lagrangian points on ε axis, like L_1, L_2, L_3 we're considering, the bracketed expression and ε have opposite signs. The right hand side must, therefore be negative, so:

$$\kappa > 1.$$

But this contradicts our previous requirement. Therefore we cannot find purely imaginary solutions for ω for L_1, L_2, L_3 . Therefore L_1, L_2, L_3 must be unstable.

For L_4 , however, we see that the linearized equations of motion are

$$\ddot{x} - 2\dot{y} = \frac{3}{4}x + \frac{3\sqrt{3}}{4}(1-2u)y$$

$$\dot{y} + 2\dot{x} = \frac{9}{4}y + \frac{3\sqrt{3}}{4}(1-2u)x$$

And therefore the trial solutions

$$x = Ae^{\omega t}$$

$$y = Be^{\omega t}$$

gives:

$$\omega^4 + \omega^2 + \frac{27}{4}u(1-u) = 0$$

Because

$$\omega_1^2 \omega_2^2 = \frac{27}{4}u(1-u) > 0$$

The possible real roots have the same sign, furthermore since:

$$\omega_1^2 + \omega_2^2 = -1 < 0$$

Both roots ω_i^2 must be negative. Since we require that

$$\omega^4 + \omega^2 + \frac{27}{4} u(1-u) = 0 \text{ has real roots and then } \omega \text{ is imaginary}$$

or:

$$27u^2 - 27u + 1 > 0$$

This happens when

$$u < \frac{1}{2} - \sqrt{\frac{23}{108}} \approx .0385$$

So L_4 and L_5 have stable orbits when $u < .0385$.

Appendix E: DF of Three-Body Problem in Rotating Inertial Frame

Equations of Motion in Rotating Inertial Frame

$$\vec{F} = \begin{pmatrix} \dot{\xi} \\ \dot{p} \\ \dot{q} \\ \dot{\eta} \end{pmatrix} = \begin{pmatrix} p \\ 2q + \xi - (1-\mu)(\xi + \mu) \left[(\xi + \mu)^2 + \eta^2 \right]^{-3/2} \\ - \mu (\xi - 1 + \mu) \left[(\xi - 1 + \mu)^2 + \eta^2 \right]^{-3/2} \\ - 2p + \eta - \mu(1-\mu) \left[(\xi + \mu)^2 + \eta^2 \right]^{-3/2} \\ - \mu \mu \left[(\xi - 1 + \mu)^2 + \eta^2 \right]^{-3/2} \end{pmatrix} q$$

$$\begin{pmatrix} 0 & 2 & -1 & 0 \\ 0 & \frac{\partial^2 f}{\partial x^2} & 0 & \frac{\partial^2 f}{\partial x \partial y} \\ -1 & 0 & 0 & -2 \\ 0 & \frac{\partial^2 f}{\partial x \partial y} & 0 & \frac{\partial^2 f}{\partial y^2} \end{pmatrix}$$

=

$$\begin{pmatrix} \frac{\partial^2 f}{\partial x^2} & \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial y^2} \\ \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial x^2} & \frac{\partial^2 f}{\partial x^2} & \frac{\partial^2 f}{\partial x \partial y} \\ \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial x^2} & \frac{\partial^2 f}{\partial x \partial y} \\ \frac{\partial^2 f}{\partial y^2} & \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial x \partial y} & \frac{\partial^2 f}{\partial y^2} \end{pmatrix}$$

$\vec{Df} =$

$$\dot{p} = 2q + \varepsilon - (1-\mu)(\varepsilon + \mu) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-3/2} - \mu(\varepsilon - 1 + \mu) \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-3/2}$$

$$\frac{\partial \dot{p}}{\partial \varepsilon} = 1 - \left\{ (1-\mu)(1) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-3/2} + (1-\mu)(\varepsilon + \mu) \left(-\frac{3}{2}\right) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-5/2} \right\} \\ - \left\{ \mu(1) \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-3/2} + \mu(\varepsilon - 1 + \mu) \left(-\frac{3}{2}\right) \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-5/2} \right\}$$

$$\frac{\partial \dot{p}}{\partial v} = -(1-\mu)(\varepsilon + \mu) \left(-\frac{3}{2}\right) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-5/2} (2v) - \mu(\varepsilon - 1 + \mu) \left(-\frac{3}{2}\right) \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-5/2} (2v)$$

$$\dot{q} = -2p + v - \mu(1-\mu) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-3/2} - \mu \mu \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-3/2}$$

$$\frac{\partial \dot{q}}{\partial \varepsilon} = -\mu(1-\mu) \left(-\frac{3}{2}\right) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-5/2} 2(\varepsilon + \mu) - \mu \mu \left(-\frac{3}{2}\right) \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-5/2} 2(\varepsilon - 1 + \mu)$$

$$\frac{\partial \dot{q}}{\partial v} = 1 - \left\{ (1-\mu) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-3/2} + \mu(1-\mu) \left(-\frac{3}{2}\right) \left[(\varepsilon + \mu)^2 + v^2 \right]^{-5/2} 2v \right\}$$

$$- \left\{ \mu \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-3/2} + \mu \mu \left(-\frac{3}{2}\right) \left[(\varepsilon - 1 + \mu)^2 + v^2 \right]^{-5/2} 2v \right\}$$